# Development and use of thermistor sensors in a Rosemount 102 housing

12<sup>th</sup> November 2020 Hannah Price, FAAM Instrument Scientist hannah.price@faam.ac.uk



#### Measuring air temperature at FAAM





- FAAM has a team of instrument scientists looking after a suite of "core" instruments
- Data from these instruments is available to anyone using the aircraft

#### Measuring air temperature at FAAM





- Indicated air temperature measured inside two Rosemount 102 housings
- Combined with pressure measurements and recovery factor to produce static air temperature

#### Measuring air temperature at FAAM





Non-de-iced housing is smaller, with smaller inlet

De-iced housing has optional heater, turned on manually in icing conditions



#### Present

Indicated temperature

PRT sensors calibrated regularly at National Physical Laboratory

Mach

Comes from static and dynamic RVSM pressure measurements

Recovery factor

Constant recovery factor for each housing measured via speed runs



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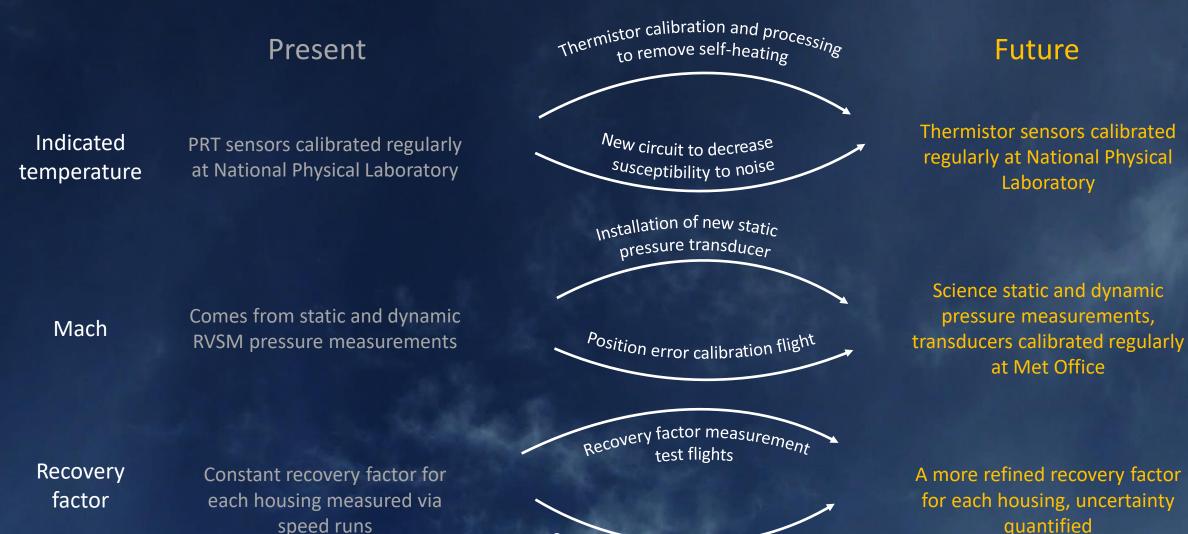
#### **Future**

Thermistor sensors calibrated regularly at National Physical Laboratory

Science static and dynamic pressure measurements, transducers calibrated regularly at Met Office

A more refined recovery factor for each housing, uncertainty quantified





Conversations with Collins Aerospace



#### Present

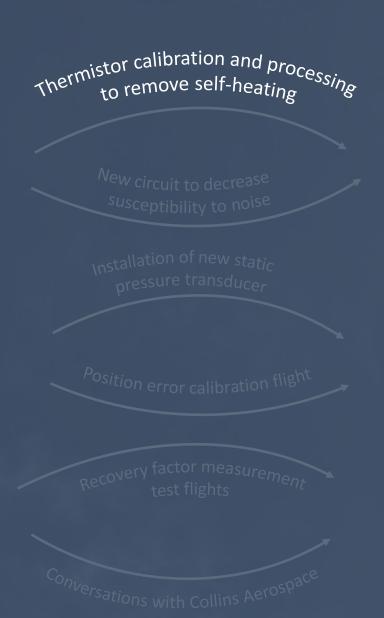
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### Measuring air temperature: PRT sensors









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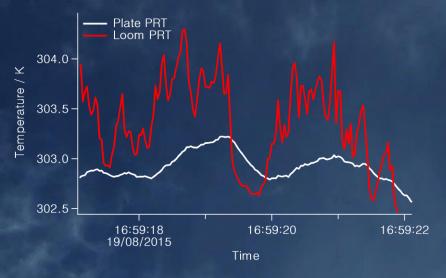






- Original sensors, "looms", were thin platinum wire wound round mica support
- Fast response, but complex due to dual time constant
- Looms are fragile and can drift / fail suddenly
- Out of production, expensive to get one made
- Commercial thin-film PRTs can be mounted in an old holder
- Low-cost, robust and minimal calibration drift
- BUT slower (~1s) time response than loom

#### In-flight plate vs loom temperature response



#### Measuring air temperature: thermistors

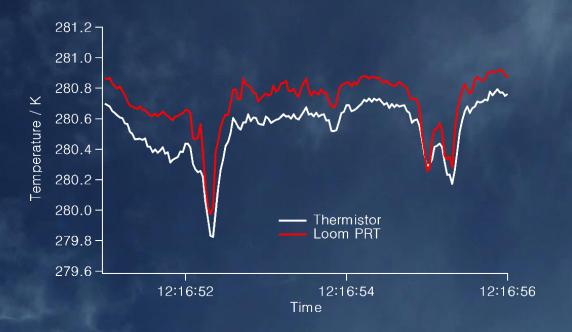


- Commercially available (GE NTC Type FP07 Fastip Thermoprobes) fast response glass bead thermistors mounted instead of PRT in old holder
- Two of type FP07DA103N, 10 k $\Omega$  at 25 °C, and two of type FP07DB154N, 150 k $\Omega$  at 25 °C
  - → Focussing here on FP07DA103N
- Initial tests indicate that response may be as fast as loom sensors, with minimal drift
- Thermistors currently susceptible to electronic noise

New circuit being designed







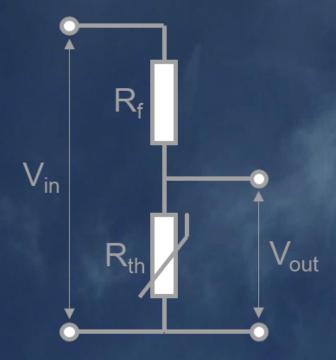
#### Calibrating FAAM's thermistors



- Thermistors self-heat
  - > needs to be handled carefully in calibration and processing
- Dissipation of heat depends on flow over sensing element
- Flow is small in the calibration lab, greater in-flight, how much?
- Need to correct for self-heating in calibration lab, and self-heating inflight, without knowing the difference between flows
- Also, self-heating is temperature dependent:

$$P = K (T_{th} - T_{air})$$

Power input to thermistor = dissipation constant x self-heating



#### Calibrating thermistors in the lab



- Calibration happens in air, in chamber refreshed at 1 litre/minute
- Want to know how resistance relates to probe temperature (not air temperature)
- Need to measure self-heating as part of the calibration, in order to infer probe temperature
- To do this, calibrate at two voltages
- Reducing voltage by  $1/\sqrt{2}$  reduces the self-heating by half
- Slight differences in resistance of thermistor measured for the two applied voltages at each calibration point allows us to calculate self-heating

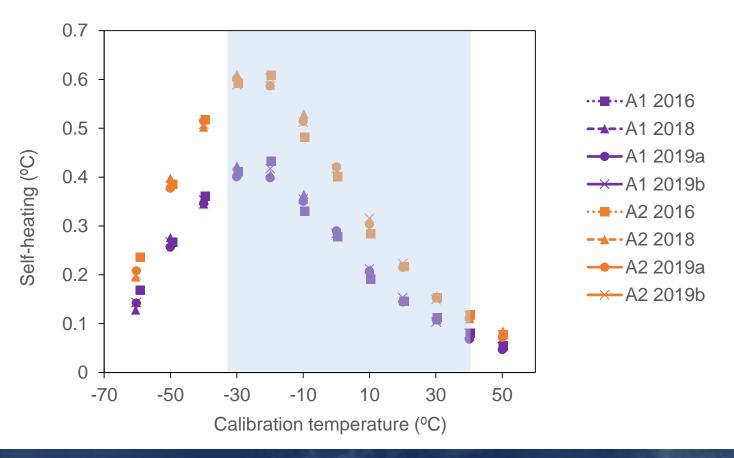


Self-heating =  $T_{th} - T_{air}$ 

#### Calibrating thermistors in the lab



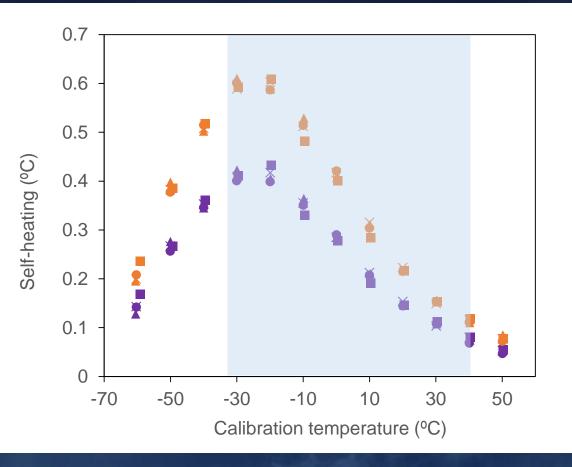
- Focussing on our two FP07DA103N thermistors, A1 and A2, each calibrated 4 times
- Both are the same type, but have different dissipation properties (glass thickness?)
- Self-heating and dissipation measurement is repeatable over 3 years



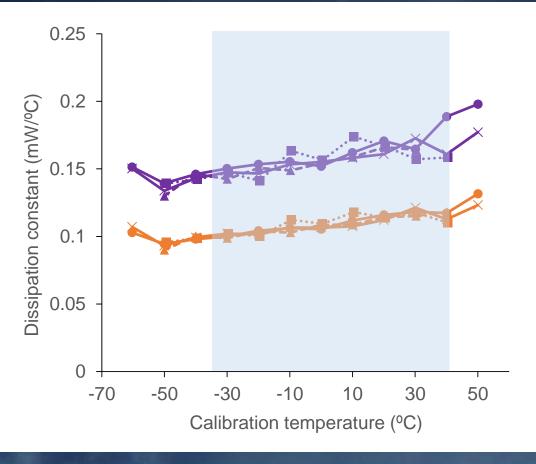
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#### Using thermistors in flight



- Now know how thermistor temperature relates to resistance
- But how much self-heating is there in flight?
- Flow through housings will be greater in flight no data (?) for how much

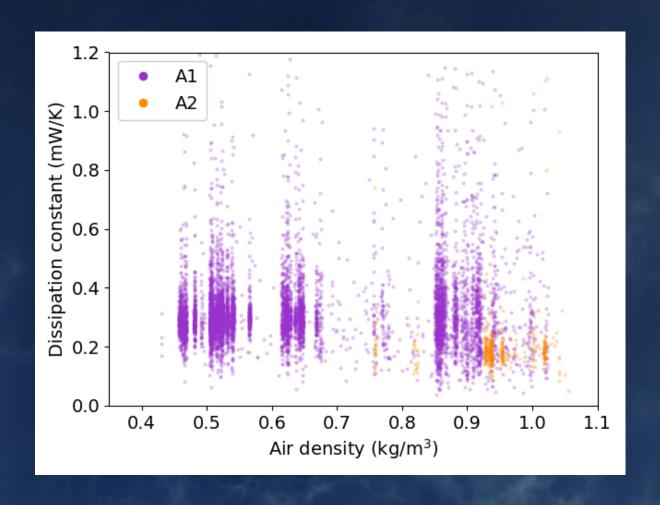
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  - Thermistors flown on 82 flights with applied voltage switched every 5 s (5 V and 5/ $\sqrt{2}$  V)
    - >100,000 measures of self-heating (along with knowing power input to thermistor)
      - >100,000 measures of dissipation constant
  - Tricky to analyse with temperature changing, so filtering required
  - Now need to look for any dependencies of dissipation on measured quantities

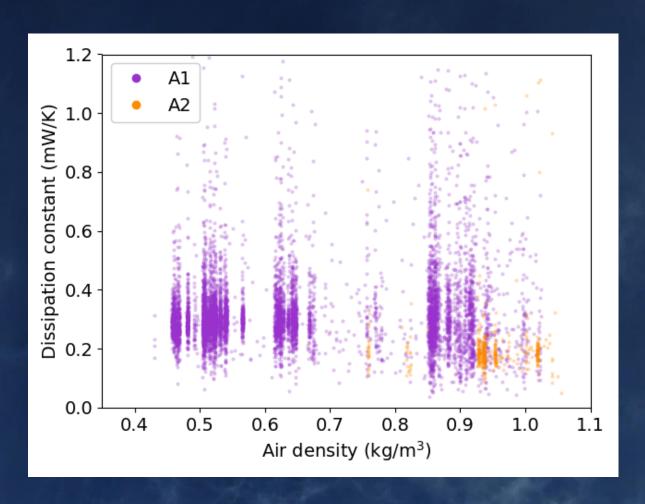


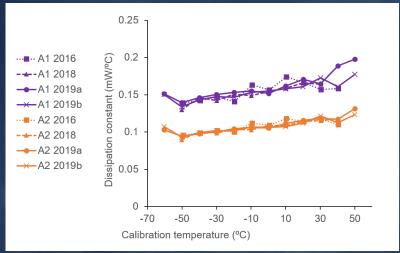
Example: does dissipation vary with air density?





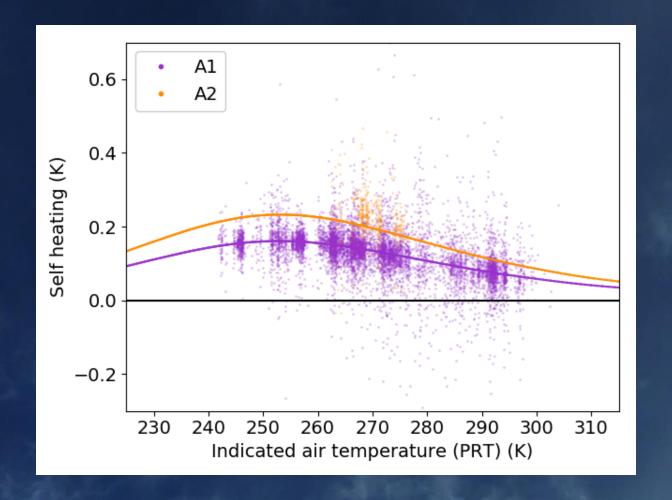
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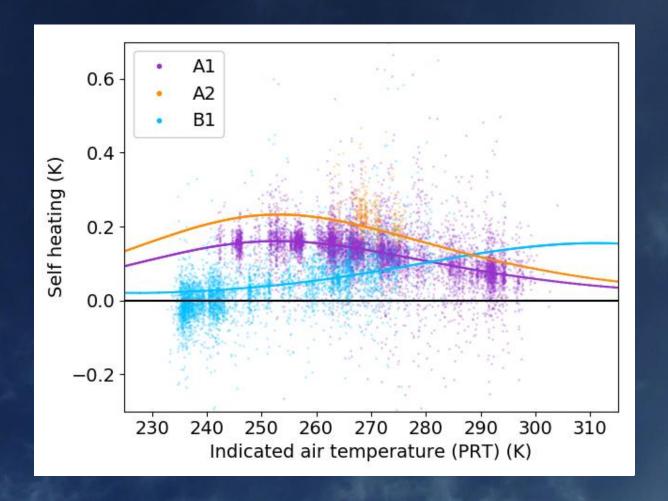


- Sensor to sensor variation exists in flight
- Lines shown are for  $k_{flight} = k_{lab} \times 1.97$



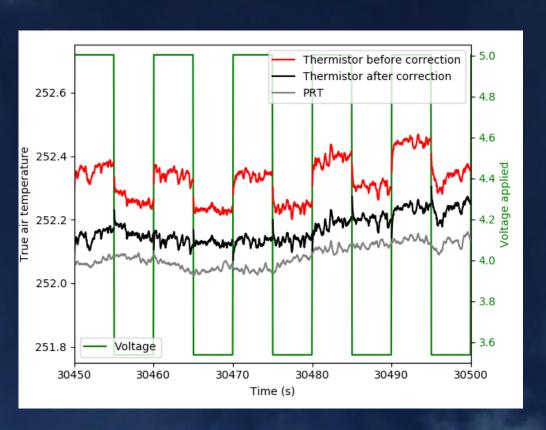


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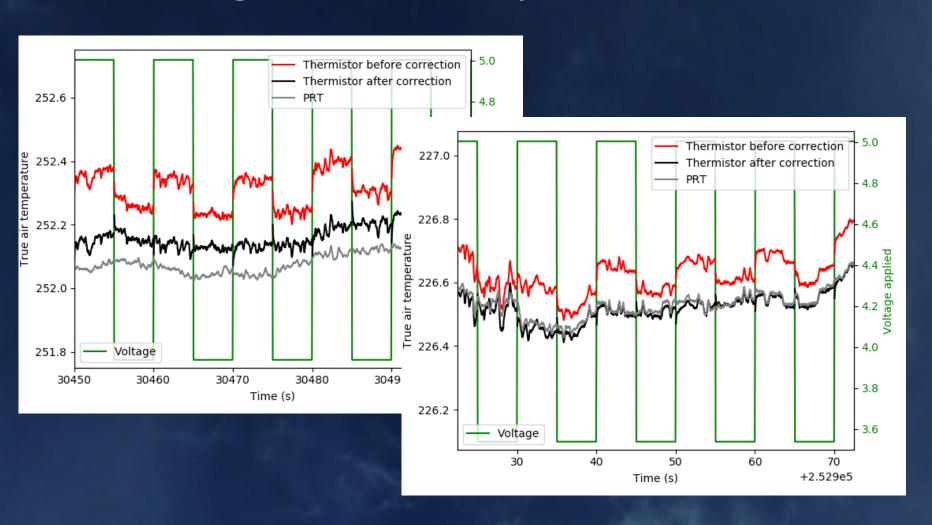
#### Measuring static air temperature: thermistors





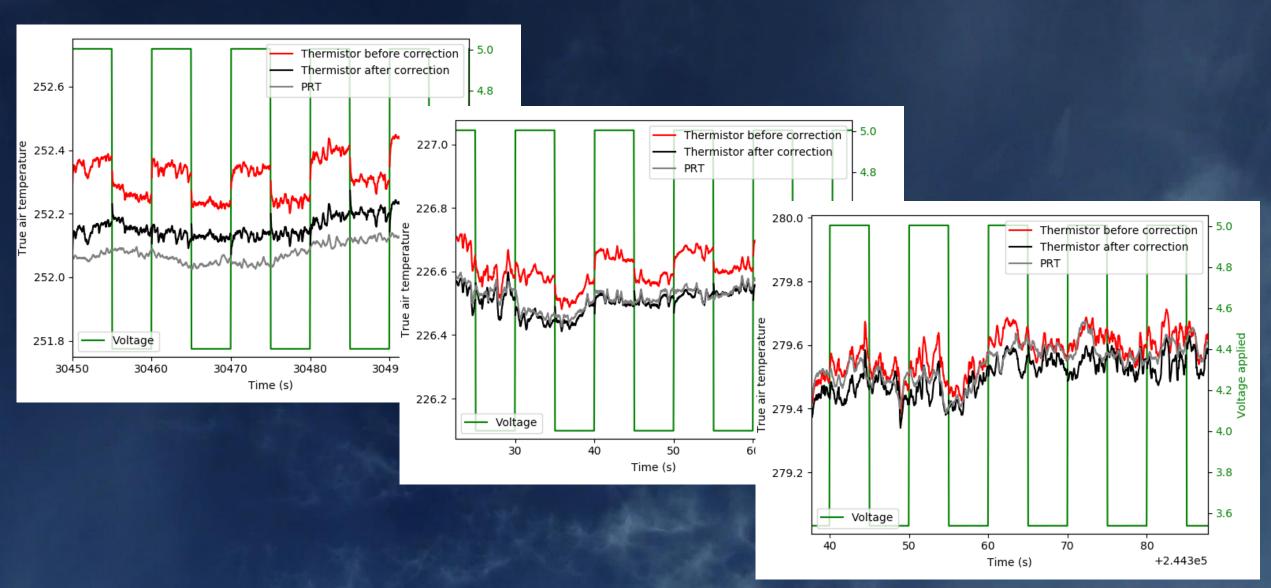
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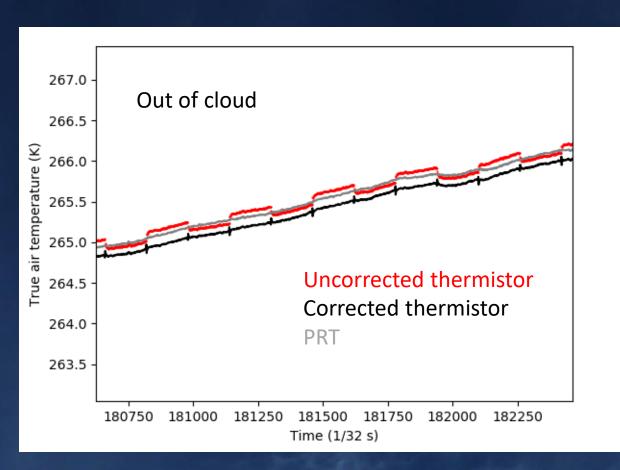


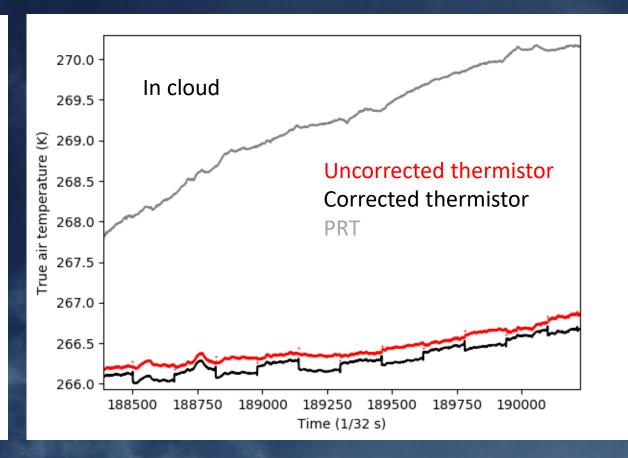


#### What about when the sensor gets wet?



Correction doesn't work when sensor is wet→ Different dissipation properties

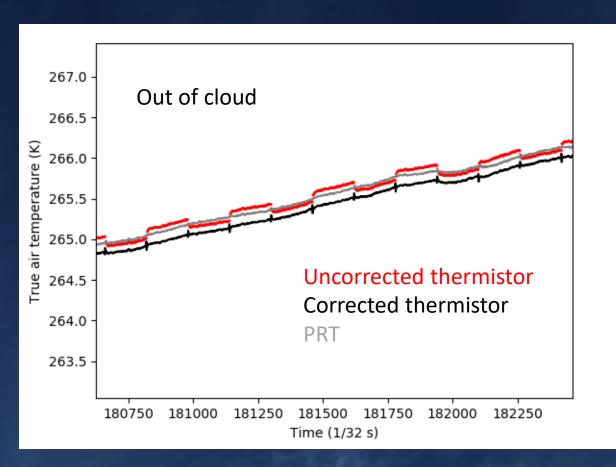


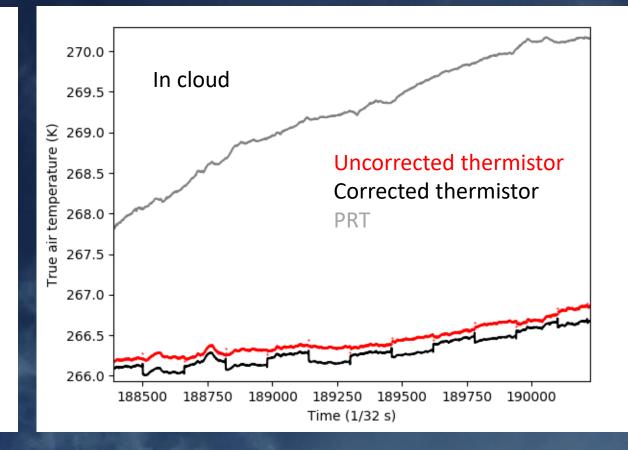


#### ...now we can verify that a sensor isn't wetted



- Correction makes things worse, so this tells us sensor is wet
- So also tells us when sensor isn't wet
- We now know when we can trust temperature measurements in cloud
- Could investigate further correlation between LWC and dissipation?





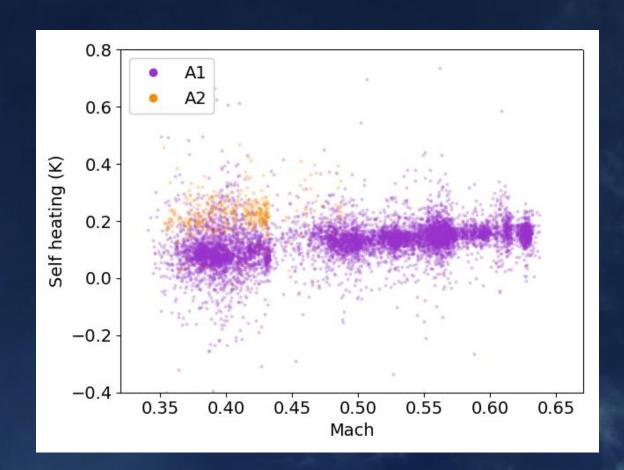
#### Summary

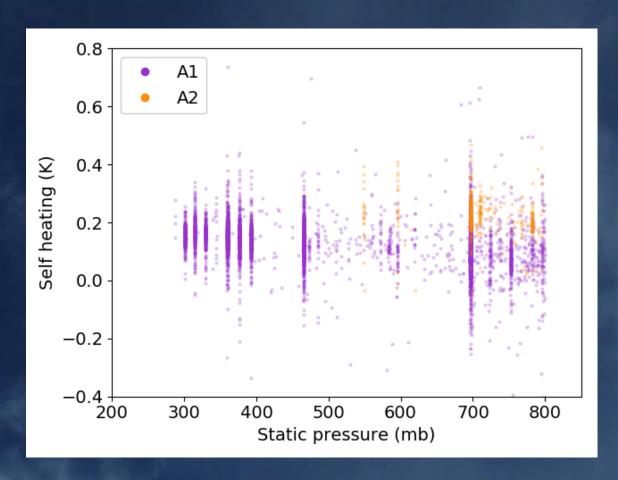
- New thermistor sensors provide a good alternative to original PRTs, flown on many flights
- Self-heating means calibration and processing isn't straightforward
  - → by switching voltages, we can measure self-heating and dissipation
- Dissipation properties evident from voltage switching in-flight tells us if sensor is wetted
  - → we can now know if a sensor is successfully measuring in-cloud air temperature



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Time response ↔ spatial resolution

Sensor needs low thermal mass – delicate!
Sensor also shouldn't get wet

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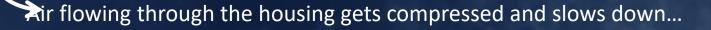


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Oh, that makes it warm up. By up to 15°C! We wanted to know its temperature!



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Rir flowing through the housing gets compressed and slows down...

Oh, that makes it warm up. By up to 15°C! We wanted to know its temperature!

Have to characterise how it warms up

$$T_i = 1 + \left(r \times \frac{\gamma - 1}{2} \times M^2\right)$$

Recovery factor of housing

Mach, comes from static and dynamic pressure

#### Measuring static air temperature

Time response ↔ spatial resolution

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## Getting temperature right

depends on getting pressure

oh, that makes it warm up. By up to 1 °CLWe wanted to know its temperature!

Have to characterise how it warms up

Temperature of the sensor

$$T_i = 1 + \left(r \times \frac{\gamma - 1}{2} \times M^2\right)$$

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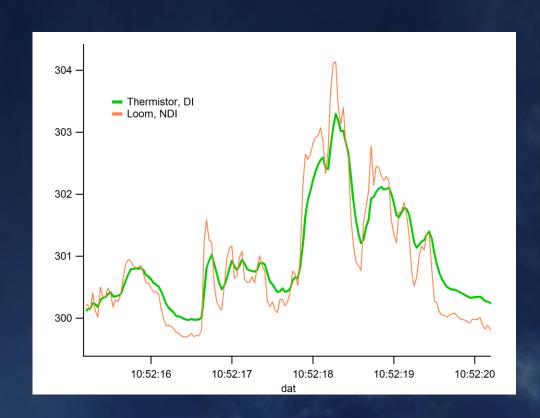
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Temperature of the sensor What we want to know 
$$\frac{T_i}{T_s} = 1 + \left(r \times \frac{\gamma - 1}{2} \times M^2\right)$$

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### Thermistors – example flight data 2019



Magnitude of variations in thermistor temperature less than for loom. Peaks and troughs occur at the same time though.

- Is the thermistor sensor slower to respond to changes in temperature?
- Does the loom overshoot?
- Is the thermistor signal overdamped by the sampling electronics?

### Measuring static air temperature: de-icing



### Measuring static air temperature: de-icing



### Measuring static air temperature: de-icing



- Housings can clog up with ice
- De-iced housing has optional heater
- More changing the temperature of the thermometer!
- Can be corrected for, but sensor wetting still an issue in both housings

